

Application to Linear Systems:

Given the system:

$$\begin{vmatrix} \dot{\underline{x}} = A\underline{x} + B\underline{u} \\ \underline{y} = C\underline{x} + D\underline{u} \end{vmatrix} \quad (\text{MIMO})$$

with state- and inputweighting:

$$PI = \underline{x}'(t_f) S \underline{x}(t_f) + \int_0^{t_f} \left\{ \underline{x}'(t) \cdot Q \underline{x}(t) + \underline{u}'(t) R \underline{u}(t) \right\} dt \stackrel{!}{=} \min_{\underline{y}(t)}$$

where: $S \geq \phi$; $Q \geq \phi$; $R > \phi$

Assume: $t_f = \underline{\text{fixed}}$; $\underline{x}(t_f) = \underline{\text{variable}}$;
 $\underline{u}(t) = \underline{\text{unlimited}}$

Q : positive semidefinite (symmetric)
state-weighting matrix

R : positive definite (symmetric)
input-weighting matrix

S : positive semidefinite final-value-weighting matrix

We can build the Hamiltonian:

$$H(\underline{x}, \underline{u}, \underline{\psi}, t) = \underline{x}' Q \underline{x} + \underline{u}' R \underline{u}$$

$$+ \underline{\psi}' (\underline{A} \underline{x} + \underline{B} \underline{u})$$

$$= \underline{x}' Q \underline{x} + \underline{\psi}' \underline{A} \underline{x} + \underline{\psi}' \underline{B} \underline{u} + \underline{u}' R \underline{u}$$

$$\Rightarrow \dot{\underline{\psi}} = - \frac{\partial H}{\partial \underline{x}} = - 2Q \underline{x} - \underline{A}' \underline{\psi}$$

where: $\underline{\psi}(t_s) = 2S \underline{x}(t_f)$

$$\frac{\partial H}{\partial \underline{u}} = \phi = 2R \underline{u}_{opt} + \underline{B}' \underline{\psi}$$

$$\rightarrow \boxed{\underline{u}_{opt}(t) = - \frac{1}{2} R^{-1} \underline{B}' \underline{\psi}}$$

$$\dot{\underline{x}} = \underline{A} \underline{x} + \underline{B} \underline{u}_{opt} = \underline{A} \underline{x} - \frac{1}{2} \underline{B} \underline{R}^{-1} \underline{B}' \underline{\psi}$$

\Rightarrow We must solve the following boundary value problem:

$$\begin{bmatrix} \dot{\underline{x}} \\ \dot{\underline{\psi}} \end{bmatrix} = \begin{bmatrix} \underline{A} & -\frac{1}{2} \underline{B} \underline{R}^{-1} \underline{B}' \\ -2Q & -\underline{A}' \end{bmatrix} \cdot \begin{bmatrix} \underline{x} \\ \underline{\psi} \end{bmatrix}$$

$$\left| \begin{array}{l} \underline{x}(\phi) = \underline{x}_0 \\ \underline{\psi}(t_f) = 2S \underline{x}(t_f) \end{array} \right|$$

We can simplify this a little bit by making:

$$\tilde{\Psi}(t) = \frac{1}{2} \cdot \underline{\Psi}(t)$$

$$\Leftrightarrow \underline{\Psi}(t) = 2 \cdot \tilde{\Psi}(t)$$

$$\Rightarrow \dot{\underline{\Psi}}(t) = 2 \cdot \dot{\tilde{\Psi}}(t)$$

$$\Rightarrow \begin{bmatrix} \dot{x}_1 \\ \dot{x}_2 \end{bmatrix} = \begin{bmatrix} A & -BR^{-1}B' \\ -Q & -A' \end{bmatrix} \cdot \begin{bmatrix} x_1 \\ x_2 \end{bmatrix}$$

$$\left| \begin{array}{l} x(\phi) = x_0 \\ \tilde{\Psi}(t_f) = S \underline{x}(t_f) \end{array} \right|$$

This $(2n \times 2n)$ "system" matrix is called the Hamiltonian matrix.

Solution:

We write for $\tilde{\Psi}(t)$:

$$\boxed{\tilde{\Psi}(t) = P(t) \cdot \underline{x}(t)}$$

As nothing was said about

$P(t)$, this is obviously possible.

$$\Rightarrow \dot{\psi}(t_f) = P(t_f) \cdot x(t_f) \equiv S x(t_f)$$

$$\Rightarrow \boxed{P(t_f) \equiv S}$$

$$\dot{x} = Ax - BR^{-1}B' \dot{\psi}$$

$$= Ax - BR^{-1}B' P_x$$

$$\Rightarrow \dot{x} = [A - BR^{-1}B' P] x$$

$$\dot{\psi} = -Qx - A' \dot{\psi}$$

$$= -Qx - A' P_x$$

$$\Rightarrow \dot{\psi} = -[Q + A' P] x$$

$$\text{but: } \dot{\psi} = P_x$$

$$\Rightarrow \dot{\psi} = (P_x)^\bullet = \dot{P}_x + P \dot{x}$$

$$-[Q + A' P] x = \dot{P}_x + P[A - BR^{-1}B' P] x$$

$$\Rightarrow \boxed{-\dot{P} = PA + A' P + Q - PBR^{-1}B' P}$$

This is called the Riccati
Differential Equation.

⇒ We have decomposed the
($2n$) - boundary value problem
into two initial value
problems, one of size (n), the
other of size ($n \times n$).

Recipe: Integrate the Riccati
equation backward in time
from $t_f \rightarrow \phi$:

$$|\dot{P} = PBR^{-1}B'P - PA - A'P - Q|$$

$$P(t_f) = S$$

$$\Rightarrow P(t)$$

Then plug the previously found
 $P(t)$ into the system equations
and integrate them forward
in time from $\phi \rightarrow t_f$:

$$\left| \dot{\underline{x}} = (A - BR^{-1}B'P(t)) \underline{x} \right| \\ \underline{x}(0) = \underline{x}_0$$

$$\Rightarrow \underline{x}(t)$$

Notice:

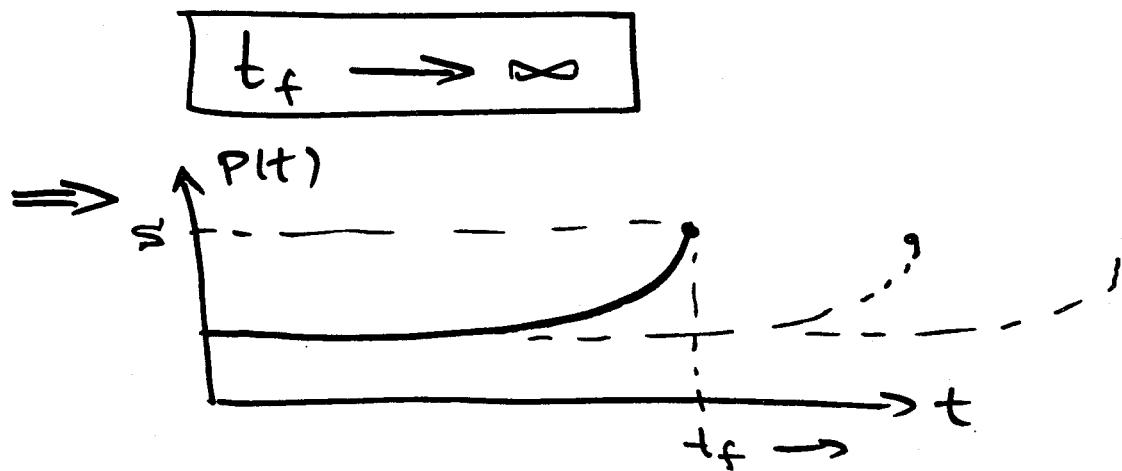
$$\begin{aligned} \underline{u}_{opt}(t) &= -\frac{1}{2} R^{-1} B' \underline{\psi}(t) \\ &= -R^{-1} B' \underline{\tilde{\psi}}(t) \\ \Rightarrow \underline{u}_{opt}(t) &= -\underbrace{R^{-1} B' P(t)}_{K(t)} \cdot \underline{x}(t) \end{aligned}$$

$$\Rightarrow \boxed{\underline{u}_{opt}(t) = \underline{r}(t) - K(t) \cdot \underline{x}(t)}$$

where: $\boxed{K(t) = R^{-1} B' P(t)}$

The optimal solution can be realized as a time-variant state-feedback.

Simplification :



$\Rightarrow P(t)$ becomes constant.

$$\Rightarrow \dot{P} = \phi$$

$$\Rightarrow PA + A'P + Q - PBR^{-1}B'P = \phi$$

\Rightarrow Algebraic Matrix - Riccati Equation.

$$\Rightarrow K = R^{-1}B'P$$

$$\Rightarrow \underline{u}_{\text{opt}}(t) = \underline{r}(t) - K \cdot \underline{x}(t)$$

is a state - feedback .

- The "algebraic Matrix - Riccati Equation" is actually a set of n^2 nonlinear equations.

- This set of equations has many solutions, but only one leads to a stable feedback system.

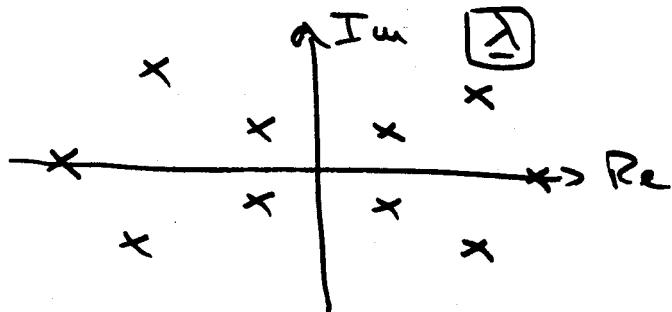
Algorithm: (without proof)

- ① Check controllability. If not completely controllable, input-decouple the uncontrollable modes first.
 - ② Compute the Hamiltonian matrix:
- $$H = \begin{bmatrix} A & -BR^{-1}B' \\ -Q & -A' \end{bmatrix}$$
- ③ Compute the spectral decomposition of H :

$$[V, \Lambda] = \text{Eig}(H)$$

The eigenvalues of the Hamiltonian are not only symmetric to the real axis, but also to the imaginary axis.

If the system is controllable
 $\Rightarrow H$ has no eigenvalues on
 the imaginary axis. E.g.:



might be a possible set of eigenvalues of the Hamiltonian.

\Rightarrow There are exactly n eigenvalues with negative real part.

- ④ Take the subset of the Right Model matrix that relates to eigenvalues with negative real part :

$$\hat{V} = \begin{matrix} n \\ & & \\ & & 2n \end{matrix}$$

- ⑤ Cut \hat{V} into an upper and a lower portion :

$$\hat{V} = \begin{bmatrix} V_1 \\ V_2 \end{bmatrix} = \begin{array}{|c|c|} \hline & n \\ V_1 & \hline \hline & n \\ V_2 & \hline \end{array}$$

- ⑥ Compute the solution to the algebraic Riccati equation :

$$P = V_2 V_1^{-1}$$

(Notice: Any other n eigenvectors would have given us another solution, but this is the one and only solution that leads to a stable closed-loop system.)

$$⑦ K = R^{-1} B' P$$

is the desired state feedback.

This algorithm is numerically very benign (much better than pole placement).

In Matlab :

```
H = [A, -(B/R)*B'; -Q, -A'];
[V, L] = eig(H);
k = 0;
for i = 1:2*n,
    if real(L(i,i)) < 0 then
        k = k + 1;
        V(:,k) = V(:,i);
    end
end
V1 = V(1:n, 1:n);
V2 = V(n+1:2*n, 1:n);
P = V2 / V1;
K = (R \ (B')) * P;
```

This algorithm also exists as a preprogrammed Matlab function:

$K = lqr(A, B, Q, R)$

↑ linear quadratic (Gaussian)
regulator

Disadvantages: LQG has a tendency of placing poles too close to each other \rightarrow large sensitivity to parameter changes.

Solution: There meanwhile exist techniques to individually influence pole locations in the Riccati design \rightarrow mixture between Riccati + pole placement \rightarrow probably better than any of the two alone
 \Rightarrow Hal Tharp

- LQG and PLACE can both be used for state-feedback design. \Rightarrow We can obtain output feedback e.g. by solving two LQG-problems one for the controller, and

one for the observer:

$$\begin{cases} \dot{x} = LQR(A, B, Q_c, R_c); \\ \dot{y} = LQR(A', C', Q_o, R_o); \\ \dot{z} = \dot{y}; \end{cases}$$

Output Weighting:

Sometimes, it is desirable to weight the outputs instead of the states:

$$\begin{cases} \dot{x} = Ax + Bu \\ \dot{y} = Cx \end{cases} \quad (\underline{D=0})$$

$$PI = \int_0^{\infty} \{ \dot{y}' Q \dot{y} + \dot{u}' R \dot{u} \} dt \stackrel{!}{=} \min_{\underline{y}(t)}$$

$$Q \geq 0; R > 0$$

$$\begin{aligned} \Rightarrow \dot{y}' Q \dot{y} &= (Cx)' Q (Cx) \\ &= \dot{x}' \underbrace{C' Q C}_{Q_n} \dot{x} \quad ; \quad Q_n \geq 0 \end{aligned}$$

$$\Rightarrow PI = \int_0^{\infty} \{ \dot{x}' Q_n x + \dot{u}' R \dot{u} \} dt \stackrel{!}{=} \min_{\underline{u}(t)}$$

is an equivalent problem with state weighting.

- The case $D \neq \emptyset$ works also, but is a little more tricky.

$$\begin{aligned} \underline{y}' Q \underline{y} &= (C\underline{x} + D\underline{u})' Q (C\underline{x} + D\underline{u}) \\ &= (\underline{x}' C' + \underline{u}' D') Q (C\underline{x} + D\underline{u}) \\ &= \underline{x}' C' Q C \underline{x} + \underline{x}' C' Q D \underline{u} + \underline{u}' D' Q C \underline{x} \\ &\quad + \underline{u}' D' Q D \underline{u} \end{aligned}$$

Let:
$$\begin{cases} \hat{Q} = C' Q C \\ \hat{R} = R + D' Q D \\ \hat{N} = C' Q D \end{cases}$$

$$\Rightarrow PI = \int_0^\infty \left\{ \underline{x}' \hat{Q} \underline{x} + \underline{x}' \hat{N} \underline{u} + \underline{u}' \hat{N}' \underline{x} + \underline{u}' \hat{R} \underline{u} \right\} dt$$

$$\approx PI = \int_0^\infty [\underline{x}' \quad \underline{u}'] \cdot \begin{bmatrix} \hat{Q} & \hat{N} \\ \hat{N}' & \hat{R} \end{bmatrix} \cdot \begin{bmatrix} \underline{x} \\ \underline{u} \end{bmatrix} dt = \min_{\underline{u} \in U}$$

Another variable transformation can remove the mixed terms:

$$PI = \int_{0}^{\infty} \left\{ \underline{x}' (\hat{Q} - \hat{N} \hat{R}^{-1} \hat{N}') \underline{x} + (\underline{u} + \hat{R}^{-1} \hat{N}' \underline{x})' \cdot \hat{R} (\underline{u} + \hat{R}^{-1} \hat{N}' \underline{x}) \right\} dt \stackrel{!}{=} \underset{\underline{u}}{\text{Min}}$$

(Can be verified easily by multiplying out.)

Let: $\begin{cases} Q_n = \hat{Q} - \hat{N} \hat{R}^{-1} \hat{N}' \\ \underline{u}_n = \underline{u} + \hat{R}^{-1} \hat{N}' \underline{x} \end{cases}$

$$\Rightarrow PI = \int_{0}^{\infty} \left\{ \underline{x}' Q_n \underline{x} + \underline{u}_n' \hat{R} \underline{u}_n \right\} dt \stackrel{!}{=} \underset{\underline{u}_n}{\text{Min}}$$

\Rightarrow is a performance index with state weighting and without mixed terms for a modified problem:

$$\begin{cases} \dot{\underline{x}} = A_n \underline{x} + B_n \underline{u}_n \\ \underline{y} = C_n \underline{x} + D_n \underline{u}_n \end{cases}$$

$$\begin{aligned}
 \dot{\underline{x}} &= C\underline{x} + D\underline{u} \\
 &= C\underline{x} + D\underline{u}_n - D\hat{R}^{-1}\hat{z}'\underline{x} \\
 \Rightarrow \dot{\underline{x}} &= \underbrace{[C - D\hat{R}^{-1}\hat{z}']}_{C_n} \underline{x} + \underbrace{D\underline{u}_n}_{D_n}
 \end{aligned}$$

$$\begin{aligned}
 \dot{\underline{x}} &= A\underline{x} + B\underline{u} \\
 &= A\underline{x} + B\underline{u}_n - B\hat{R}^{-1}\hat{z}'\underline{x} \\
 \Rightarrow \dot{\underline{x}} &= \underbrace{[A - B\hat{R}^{-1}\hat{z}']}_{A_n} \underline{x} + \underbrace{B\underline{u}_n}_{B_n}
 \end{aligned}$$

$$\Rightarrow \left. \begin{array}{l} \dot{\underline{x}} = A_n \underline{x} + B\underline{u}_n \\ \dot{\underline{y}} = C_n \underline{x} + D\underline{u}_n \end{array} \right\}$$

where: $\left. \begin{array}{l} A_n = A - B\hat{R}^{-1}\hat{z}' \\ C_n = C - D\hat{R}^{-1}\hat{z}' \end{array} \right\}$

$$PI = \int_0^{\infty} \{ \underline{x}' Q_n \underline{x} + \underline{u}_n' \hat{R} \underline{u}_n \} dt \stackrel{!}{=} \min_{\underline{u}_n}$$

with:

$$\left| \begin{array}{l} \hat{Q} = C' Q C \\ \hat{N} = C' Q D \\ \hat{R} = R + D' Q D \\ Q_n = \hat{Q} - \hat{N} \hat{R}^{-1} \hat{N}' \end{array} \right|$$

$$\Rightarrow \hat{H} = \begin{bmatrix} A_n & -B \hat{R}^{-1} B' \\ -Q_n & -A_n' \end{bmatrix}$$

$$\Rightarrow \dots \Rightarrow \hat{K} = B' \hat{R}^{-1} \hat{P}$$

↑ for new formulation

$$\Rightarrow \underline{u}_n = \underline{r} - \hat{K} \underline{x} \equiv \underline{u} + \hat{R}^{-1} \hat{N}' \underline{x}$$

$$\Rightarrow \underline{u} = \underline{r} - \underbrace{\left[\hat{K} + \hat{R}^{-1} \hat{N}' \right]}_{\hat{K}} \underline{x}$$

$$\boxed{\underline{u} = \underline{r} - \hat{K} \underline{x}}$$

is again a state feedback

where:

$$\boxed{\hat{K} = \hat{K} + \hat{R}^{-1} \hat{N}'}$$

↑ for old formulation

In Matlab:

Let us solve the problem:

$$\begin{cases} \dot{x} = Ax + Bu \\ y = Cx + Du \end{cases}$$

$$PI = \int_0^{\infty} [x' u'] \cdot [\hat{Q}, \hat{N}] \cdot \begin{bmatrix} x \\ u \end{bmatrix} dt \stackrel{!}{=} \min_{u(t)}$$

Algorithm:

$$\begin{cases} A_n = A - (B/\hat{R}) * \hat{N}' \\ Q_n = \hat{Q} - (\hat{N}/\hat{R}) * \hat{N}' \\ \hat{K} = lqr(A_n, B, Q_n, \hat{R}) \\ K = \hat{K} + (\hat{R}/\hat{N}') \end{cases}$$

This also exists as a pre-programmed
Matlab function:

$$K = lqr(A, B, \hat{Q}, \hat{R}, \hat{N});$$

Let us now solve the complete output weighting problem:

$$\begin{cases} \dot{x} = A \cdot x + B \cdot u \\ y = C \cdot x + D \cdot u \end{cases}$$

$$PI = \int_0^{\infty} [y' u'] \cdot \begin{bmatrix} Q & 0 \\ 0 & R \end{bmatrix} \cdot \begin{bmatrix} y \\ u \end{bmatrix} dt = \min_u(t)$$

Algorithm :

$$\begin{cases} \hat{Q} = C' * Q * C; \\ \hat{R} = R + D' * Q * D; \\ \hat{N} = C' * Q * D; \\ K = lqr(A, B, \hat{Q}, \hat{R}, \hat{N}); \end{cases}$$

This can also be obtained by:

$$\begin{cases} N = \text{zeros}(\underbrace{P}_{\# \text{ of outputs}}, \underbrace{m}_{\# \text{ of inputs}}); \\ K = lgry(A, B, C, D, Q, R, N); \end{cases}$$

Exponential Stability:

In order to guarantee a maximum settling time, we want all eigenvalues by at least a factor of α left from the imaginary axis, where:

$$G \approx \frac{4}{T_s}$$

With optimal control, this problem can also be solved very easily.

$$\begin{cases} \dot{\underline{x}} = A\underline{x} + B\underline{u} \\ \underline{y} = C\underline{x} + D\underline{u} \end{cases}$$

be a system to be controlled.

- We can look at a different system:

$$\begin{cases} \dot{\underline{x}}_n = \tilde{A}\underline{x}_n + B\underline{u} \\ \underline{y}_n = C\underline{x}_n + D\underline{u} \end{cases}$$

where: $\tilde{A} = A + \alpha I$

$$\Rightarrow \{\text{Eig}(\tilde{A})\} = \{\text{Eig}(A) + \alpha\}$$

We now design the state feedback for the new system: K

$$\Rightarrow \tilde{A}_{CL} = \tilde{A} - BK$$

is stable.

- We now apply this feedback K to our original problem:

$$\Rightarrow A_{CL} = A - BK$$

$$\begin{aligned}\Rightarrow A_{CL} &= \tilde{A} - \alpha I - BK \\ &= \tilde{A} - BK - \alpha I \\ &= \tilde{A}_{CL} - \alpha I\end{aligned}$$

$$\Rightarrow \{\text{Eig}(A_{CL})\} = \{\text{Eig}(\tilde{A}_{CL}) - \alpha\}$$

\uparrow
at least α
left from
imaginary
axis.

\uparrow
stable

Example:

Given the system:

$$\dot{\underline{x}} = \begin{bmatrix} -150 & 192 & 12 & 165 \\ 143 & -181 & -15 & -154 \\ -142 & 179 & 15 & 153 \\ -291 & 370 & 28 & 316 \end{bmatrix} \underline{x} + \begin{bmatrix} 1 \\ -1 \\ 1 \\ 2 \end{bmatrix} u$$

$$y = \begin{bmatrix} -27 & 51 & -18 & 48 \end{bmatrix} \underline{x}$$

(cf. page 232)

- We want to design an output feedback using optimal control where all controller poles are at least 4 from the imaginary axis, all observer poles are at least 8 from the imaginary axis. We use output weighting with:

$$Q = [1] \text{ and } R = [1].$$

- We already have found a model for the system:

$$\left| \begin{array}{l} \dot{\underline{y}} = \begin{bmatrix} \phi & 1 & \phi \\ \phi & \phi & 1 \\ -24 & 2 & 5 \end{bmatrix} \underline{y} + \begin{bmatrix} \phi \\ \phi \\ 1 \end{bmatrix} u \\ \underline{y} = [3 \ 3 \ \phi] \underline{y} \end{array} \right|$$

(cf. page 235)

$$\Rightarrow \begin{array}{l} [> A = [\phi, 1, \phi; \phi, \phi, 1; -24, 2, 5]; \\ [> B = [\phi; \phi; 1]; \\ [> C = [3, 3, \phi]; \\ [> D = \emptyset; \end{array}$$

$$[> AT = A + 4 * EYE(A);$$

$$[> Q = 1;$$

$$[> R = 1;$$

$$[> N = \emptyset;$$

$$[> K = LQRY(AT, B, C, D, Q, R, N)$$

$$\Rightarrow K = [769.4225 \quad 272.1969 \quad 34.0079]$$

$$[> EIG(A - B * K)$$

$$\Rightarrow ANS = \begin{bmatrix} -6.0042 \\ -11.0395 \\ -11.9642 \end{bmatrix} \quad \underline{\text{all real !!!}}$$

Observer design:

[> $A\bar{T} = A + 8 * EYE(A)$;

[> $H = LQRY(A\bar{T}', C', B', D', Q, R, N)$];

[> $H = H'$

$$\Rightarrow H = \begin{bmatrix} -73.7788 \\ 93.1127 \\ 312.9170 \end{bmatrix}$$

[> $EIG(A - H + C)$

$$\rightarrow ANS = \begin{bmatrix} -14.0011 \\ -19.0153 \\ -19.9854 \end{bmatrix}$$

$\Rightarrow K'$ and B are too large.

Balancing:

[> $T = SQRT(ABS(K' ./ H))$;

[> $T = DIAG(T)$;

[> $AN = T * A / T$;

[> $BN = T * B$;

[> $CN = C / T$;

$$\Rightarrow \dot{y} = \begin{bmatrix} 0 & 1.8893 & 0 \\ 0 & 0 & 5.1864 \\ -2.4507 & 0.3856 & 5 \end{bmatrix} y + \begin{bmatrix} 0 \\ 0 \\ 0.3297 \end{bmatrix} u$$

$$y = [0.9292 \ 1.7546 \ 0] y$$

is a new model.

$$\Rightarrow [> K_N = K/T]$$

$$\Rightarrow K_N = \begin{bmatrix} 238.1965 & 159.2011 & 103.1583 \end{bmatrix}$$

$$[> H_N = T \times H]$$

$$\Rightarrow H_N = \begin{bmatrix} -238.1965 \\ 159.2011 \\ 103.1583 \end{bmatrix}$$

\Rightarrow We have requested too much.

Looking at the closed loop eigenvalues, we realize that LQR has placed them too far to the left.

- We repeat the design by lowering our demands:

controller poles left from -2.5
observer poles left from -5

(in the hope that LQR still places them where we want them to be).

$\Rightarrow \{ > AT = A + 2.5 * EYE(A);$
 $\{ > K = LQR(Y(AT, B, C, D, Q, R, N))$

$\Rightarrow K = [193.4091 \quad 125.3972 \quad 25.4227]$
 $\{ > EIG(A - B * K)$

$\Rightarrow ANS = \begin{bmatrix} -3.014 \\ -8.4676 \\ -8.9411 \end{bmatrix}$ NO!

\Rightarrow iterate a little ...

$\{ > AT = A + 3 * EYE(A);$
 $\{ > K = LQR(Y(AT, B, C, D, Q, R, N))$
 $\Rightarrow K = [337.1582 \quad 162.2872 \quad 28.0144]$

$\{ > EIG(A - B * K)$
 $\Rightarrow ANS = \begin{bmatrix} -4.008 \\ -9.4554 \\ -9.9509 \end{bmatrix} \leftarrow$ YES!

$\{ > AT = A + 6 * EYE(A);$
 $\{ > H = LQR(Y(AT', C', B', D', Q, R, N));$
 $\{ > H = H'$
 $\Rightarrow H = \begin{bmatrix} -31.1798 \\ 46.5142 \\ 214.1975 \end{bmatrix}$

$\{ > EIG(A - H * C)$
 $\Rightarrow ANS = \begin{bmatrix} -18.0019 \\ -15.0232 \\ -15.9784 \end{bmatrix} \leftarrow$ iterate a little ...

- 300 -

$$[> AT = A + S * EYE(A);$$

$$[> H = LQRY(AT', C', B', D', Q, R, N);$$

$$[> H = H'$$

$$\Rightarrow H = \begin{bmatrix} -17.8802 \\ 31.2152 \\ 168.8428 \end{bmatrix}$$

$$[> EIG(A - H * C)$$

$$\Rightarrow A_N S = \begin{bmatrix} -8.0026 \\ -13.0297 \\ -13.9726 \end{bmatrix} \leftarrow \underline{\text{very good!}}$$

Balancing:

$$[> T = \text{SQRT}(ABS(K' / H));$$

$$[> T = \text{DIAG}(T);$$

$$[> K_N = K / T, \quad H_N = T * H$$

$$\Rightarrow K_N = [77.6431 \quad 72.4784 \quad 68.7752]$$

$$H_N = \begin{bmatrix} -77.6431 \\ 72.4784 \\ 68.7752 \end{bmatrix}$$

This design looks okay.

The new model is:

$$\left\{ \begin{array}{l} \dot{\underline{x}} = \begin{bmatrix} 0 & 1.8702 & 0 \\ 0 & 0 & 5.7002 \\ -2.2513 & 0.3509 & 0 \end{bmatrix} \underline{x} + \begin{bmatrix} 0 \\ 0 \\ 0.4073 \end{bmatrix}, \\ y = [0.6909 \quad 1.292 \quad 0] \underline{x} \end{array} \right.$$

Verification:

$$\begin{aligned}& [> \mathbf{A}N = \mathbf{T} * \mathbf{A} / \mathbf{T}; \\& [> \mathbf{B}N = \mathbf{T} * \mathbf{B}; \\& [> \mathbf{C}N = \mathbf{C} / \mathbf{T}; \\& [> \mathbf{A}\mathbf{T} = \mathbf{A}N + 3 * \text{EYE}(\mathbf{A}N); \\& [> \mathbf{K} = \text{LQRY}(\mathbf{A}\mathbf{T}, \mathbf{B}N, \mathbf{C}N, \mathbf{D}, \mathbf{Q}, \mathbf{R}, N) \\ \Rightarrow \mathbf{K} = & \underline{\underline{[77.6431 \quad 72.4784 \quad 68.7752]}}} \end{aligned}$$

✓

$$\begin{aligned}& [> \mathbf{A}\mathbf{T} = \mathbf{A}N + 5 * \text{EYE}(\mathbf{A}N); \\& [> \mathbf{H} = \text{LQRY}(\mathbf{A}\mathbf{T}', \mathbf{C}N', \mathbf{B}N', \mathbf{D}', \mathbf{Q}, \mathbf{R}, N); \\& [> \mathbf{H} = \mathbf{H}' \\ \Rightarrow \mathbf{H} = & \underline{\underline{[-77.6431 \\ & \quad 72.4784 \\ & \quad 68.7752]}}} \end{aligned}$$

✓

From this, we can learn that obviously optimality and displacement of eigenvalues (exponential stability) are invariant to this kind of similarity transformation.